



Lunar Facility Heat Exchanger

Active Thermal Management for Industrial Processes on The Moon

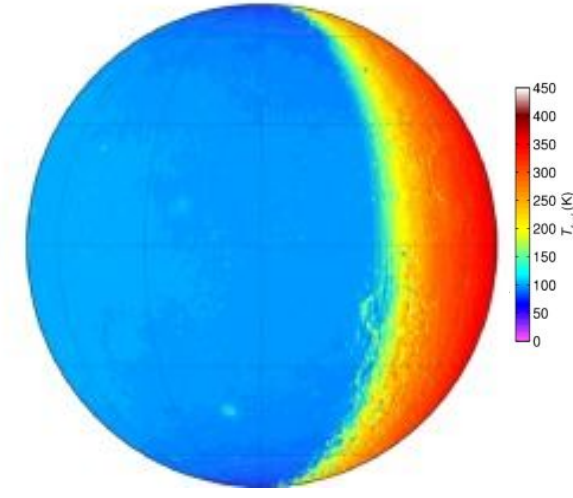
Aaron Li, Alvin St. John, Cameron Lamont, Fatin Hossain, Jessy Xu, Vinshie Chiew
Schulich School of Engineering, University of Calgary



ABSTRACT AND INTRODUCTION

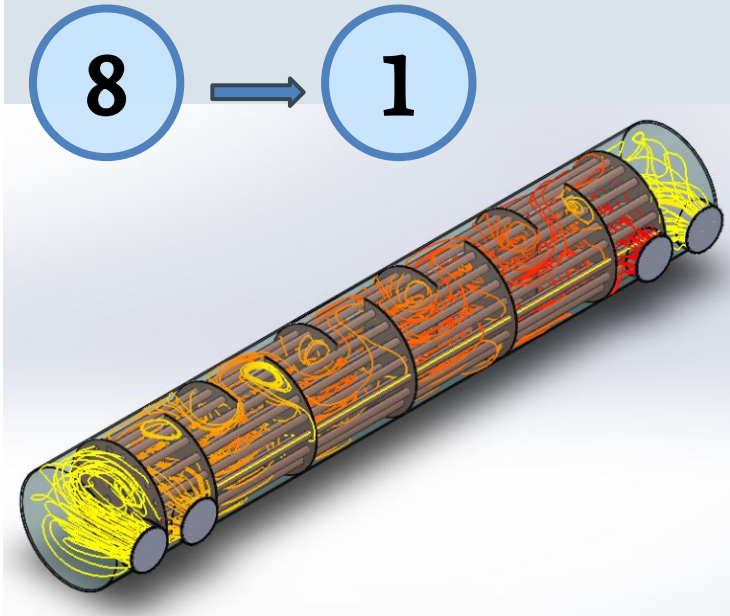
The design challenge is to investigate the technology and controls needed to dissipate heat generated from an industrial processes to a lunar environment.

The key goal is to remove 50kW of waste heat from an arbitrary facility while maintaining temperature and pressure conditions of 25C and 1 atm.



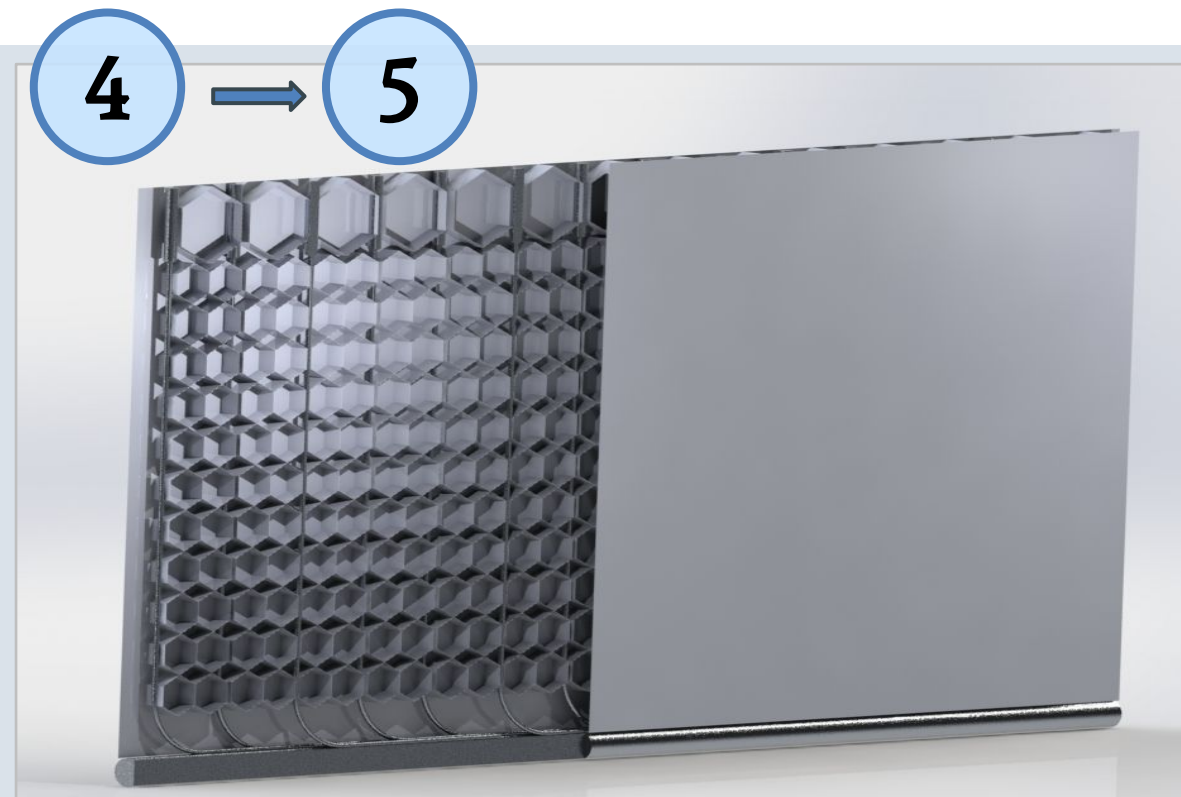
Design Constraints	Boundary Conditions
<ul style="list-style-type: none"> Weight restriction: 1000kg Power Budget: 10kW Operation: autonomous for 6-month periods (bi-annual cleaning) 	<ul style="list-style-type: none"> 14 day and night cycle Vacuum of Space – no free convection for heat expulsion Temperature Range of 50K to 450K

DISCUSSION



Facility heat exchanger:

- 50kW of waste heat is removed from the facility via a hot water loop
- Shell side contains hot water
- Tube side contains R11 refrigerant

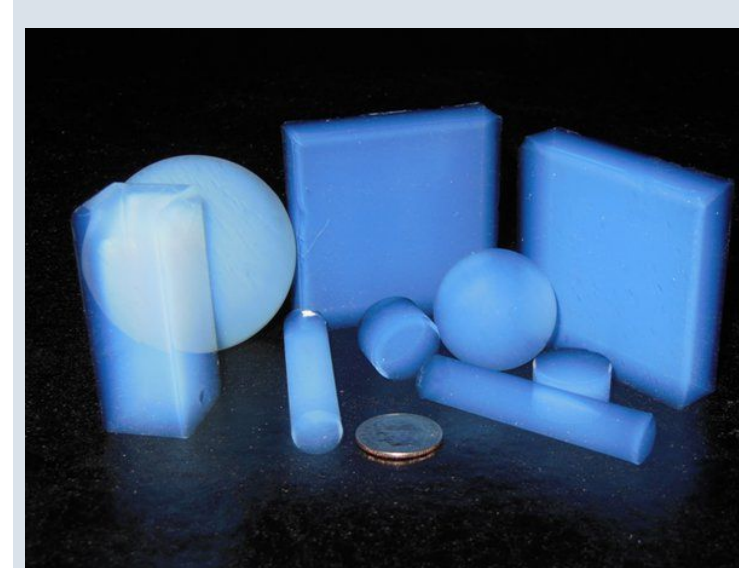


Radiators and Heat Pipes:

- Radiation is the only way to reject heat since there is not atmosphere on the moon
- Heat pipes rapidly transfer heat from the R11 to the ~400 m² of panels
- Weight is minimized by using thin hexagonal supports and optimizing the array of heat pipes

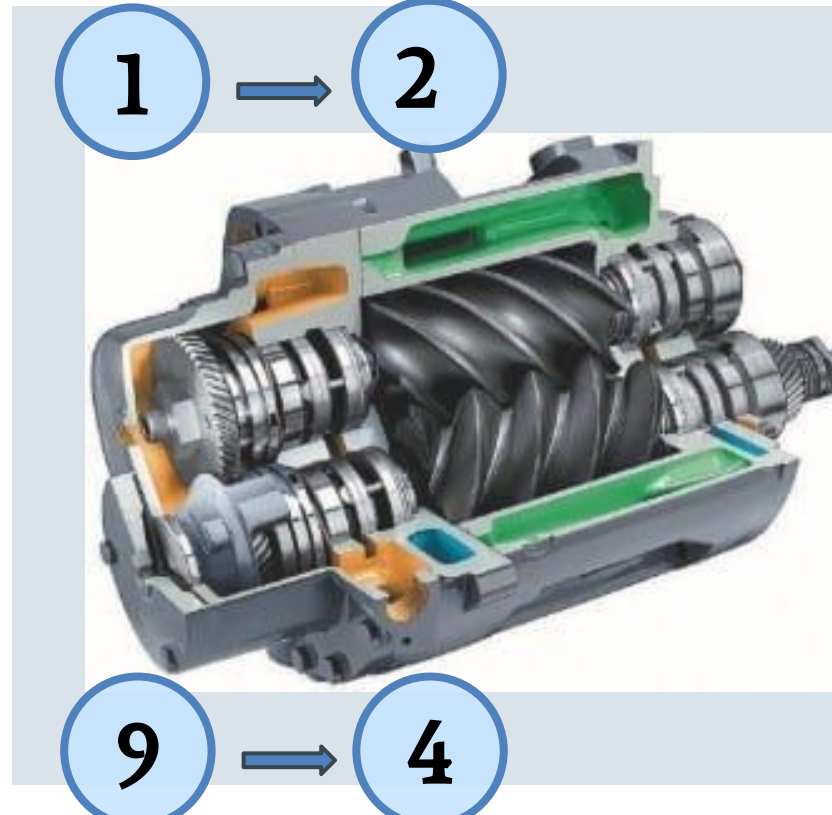
Heat Shielding:

- A shield is required to prevent our system from heating up due to the direct sunlight during day
- AZ-93 coating will be applied to the outside of the radiators to reflect away the radiation and prevent our heat pipes from becoming inefficient



Insulation:

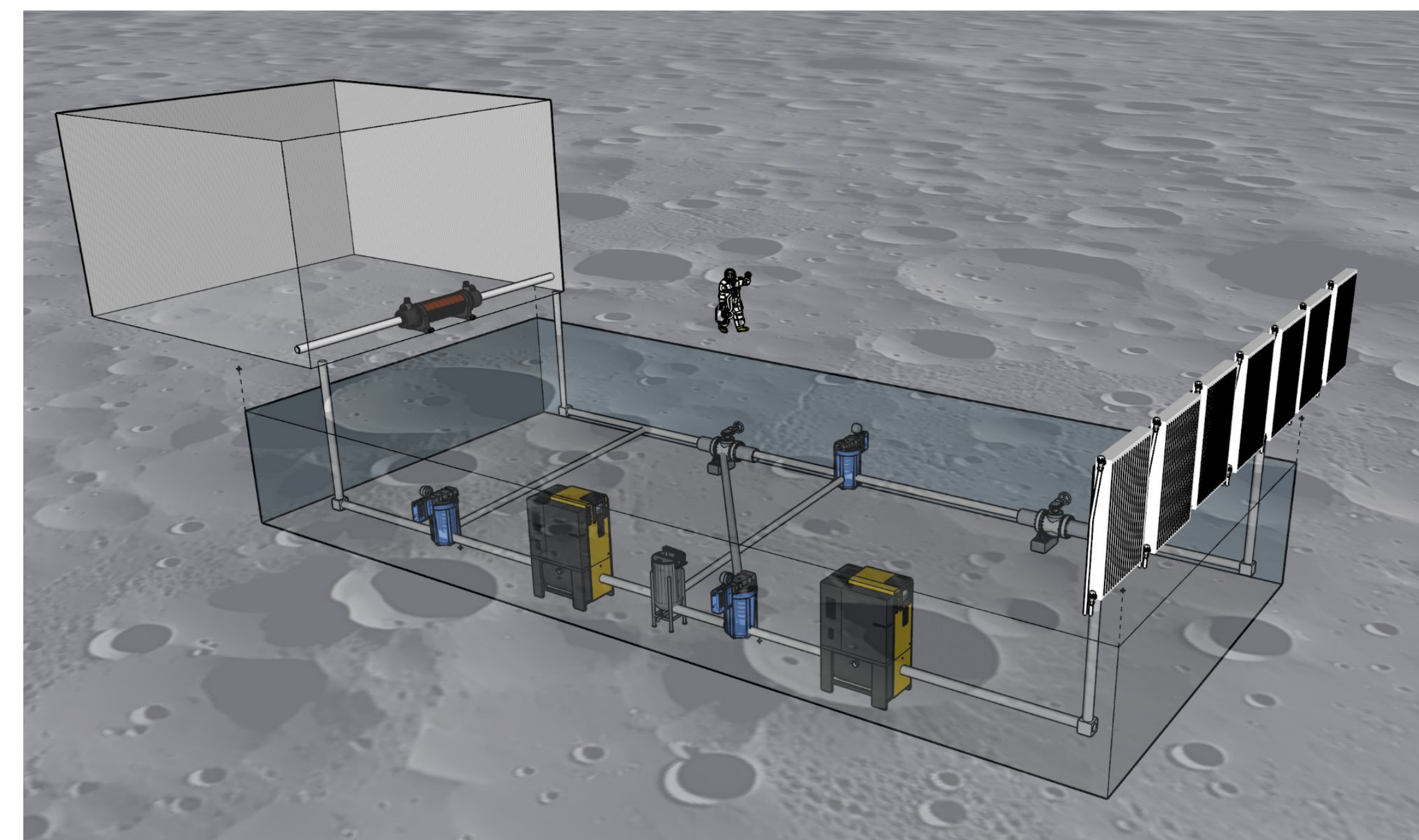
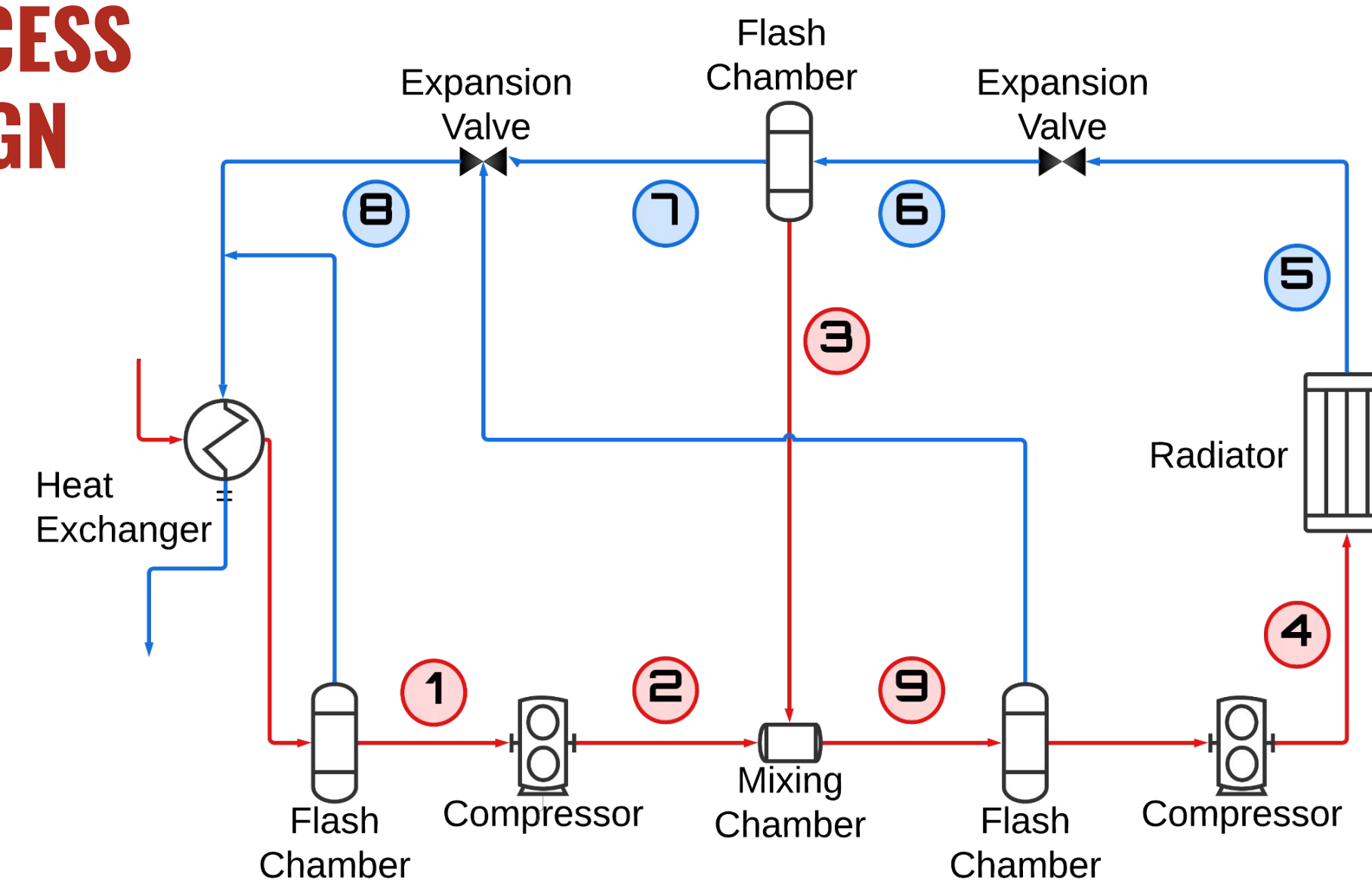
- Lunar regolith is a suitable insulator for underground components
- Coated in Cr3C2-NiCr for abrasive protection
- Aerogel used for above ground components because air is lightweight and acts as a good insulator.



Compressors and Flash Chamber:

- The multistage cycle requires the use of two compressors and a flash chamber
- Oil free rotary compressors are required because of the moon's microgravity
- The flash chamber has been selected to have efficient separation for easier maintenance.

PROCESS DESIGN



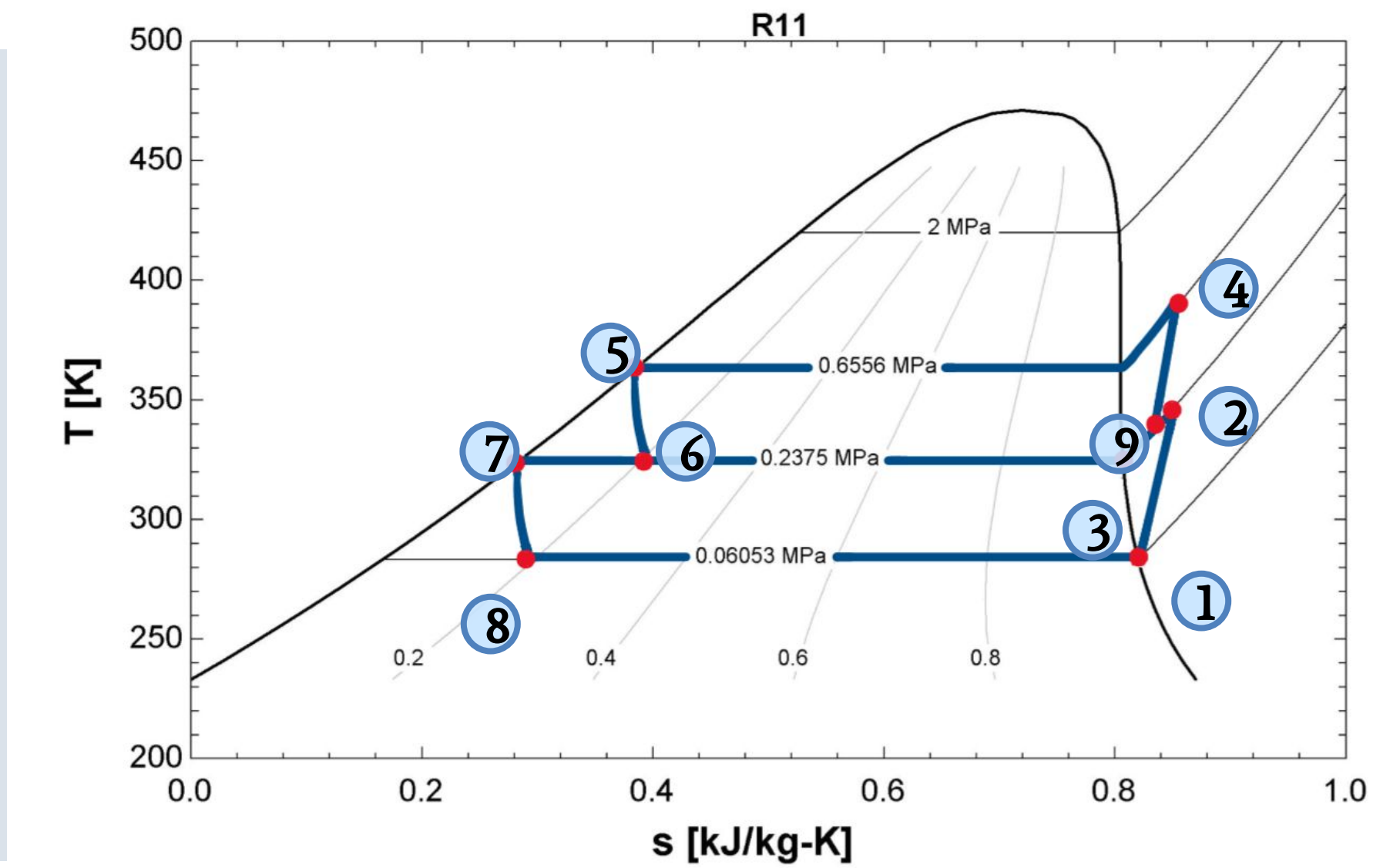
METHODS

Engineering Equation Solver (EES) and Solidworks: It has a larger database of information regarding fluids and is more efficient than performing traditional calculations ourselves.

- Fluid criteria: corrosion to metals, triple point pressure, viscosity, conductivity, molar mass
- Calculations performed for optimization: heat pipe specifications, heat exchanger weight, multistage cycle
- Solidworks: used to run simulations for verification

Thermodynamics: Using our knowledge of thermodynamics, we compared single loop, cascading, and multistage cycles.

- A constant heat load and a wide variety of external conditions necessitates for an active heat transfer system.
- Multistage adds more components and weight, but lowers the overall power budget and provides a COP of greater than 1.



MATERIALS

- Shell and tube heat exchanger:** 304 Stainless Steel with copper tubes inside
- Pipes and valves:** 304 Stainless Steel is chosen to balance weight, strength, and durability
- Heat Shield Coating:** AZ-93 White Thermal Control, Inorganic Paint minimizes the amount of solar radiation absorbed while still rejecting 89-93% of the heat
- Radiator Assembly:** Aluminum 6061-T6 is commonly used for space application
- Sensors:** Silicon Diode Sensors have high accuracy and can withstand extreme temperatures
- Working Fluid:** R11 meets our pressure, temperature, and triple point requirements. It is also less toxic than other alternatives.
- Heat Pipe Fluid:** Ammonia is commonly used in space application

RESULTS

- The shell and tube heat exchanger is 35.64kg, making it lighter than the alternative 142.5kg plate heat exchanger.
- This system requires 22kW of power at lunar noon; this exceeds the power budget in order to achieve a reasonable size and weight for the radiator panel.
- A simulation prototype was developed to demonstrate the interdependencies between each section of the system. This provides an example of a SCADA view of the system.

CONCLUSION

- Our design is able to remove 50 kW from a facility and maintain a temperature of 25C and pressures of 1atm. It will weigh about 12475 kg and has a maximum power requirement of 22kW.
- Our final design is a multistage cycle with R11 as the operating fluid. A shell and tube heat exchanger operates inside the facility, and a heat pipe radiator rejects heat into the vacuum of space via radiation.
- Next Steps:** For future improvements, we would recommend looking into optimizing the shell and tube heat exchanger and consider axial grooved wick structures to improve heat pipe efficiency.

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CONTACT US

Zac Trolley

Phone: 403-618-9237

Email: Zac.Trolley@lunarwatersupply.com

